

# STABILIZED ACTUATION OF A NOVEL NiTi SHAPE-MEMORY-ALLOY-ACTUATED FLEXIBLE STRUCTURE UNDER THERMAL LOADING

## STABILIZIRAN ODZIV NOVE NiTi ZLITINE Z OBLIKOVNIM SPOMINOM – ODZIV FLEKSIBILNE STRUKTURE NA TERMIČNO OBREMENITEV

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Shape-memory-alloy (SMA) actuated flexible structures are used in a variety of configurations in many aerospace, robotics and underwater applications. In this study, a pair of nickel-titanium (NiTi) shape-memory-alloy plates is embedded in a flexible structure. An antagonistic design with a plate geometry is selected for the NiTi SMA to achieve bidirectional flexibility. A three-point-bending test is conducted to reveal the bending strength of the NiTi plates in the martensite and austenite phases. The antagonistic NiTi SMA plates are geometrically adapted and embedded into the flexible structure, which is fabricated through additive manufacturing using thermoplastic-polyurethane (TPU) flexible filament. The stabilized actuation stages of the antagonistic NiTi SMA plates embedded in the flexible structure are examined through observation of an extended number of thermal cycles. A comparison is made by applying two different electrical-current values with a regulated high-current DC power supply. A cycling profile with a maximum, bidirectional, stabilized actuation stroke of 52 mm is obtained through 100 heating/cooling cycles for the NiTi SMA-plate-actuated flexible structure. The effects of the high and low heating/cooling cyclic periods on the stabilized actuation stroke are also investigated.

Keywords: shape-memory alloy, nickel titanium, thermal loading, cyclic loading

Zlitine z oblikovnim spominom (SMA; angl.: Shape Memory Alloys) se kot odzivne fleksibilne strukture (aktuatorji) uporabljajo na različnih področjih: v letalstvu, robotiki in podvodnih aplikacijah. Avtor v članku predstavlja študijo para nikelj-titanovih plošč (NiTi) z oblikovnim spominom, vgrajenih v fleksibilno strukturo. Izbrali so antagonističen dizajn geometrije NiTi SMA plošč, da bi dosegli dvosmerno fleksibilnost. Izvedli so tritočkovni upogibni preizkus, da bi ugotovili upogibno trdnost martenzitne in austenitne faze NiTi plošč. NiTi SMA plošči so pritrčili in vgradili v fleksibilno strukturo, ki so jo izdelali s pomočjo dodajalnega postopka (AD; angl.: additive manufacturing). Pri tem so uporabili fleksibilna vlakna iz termoplastičnega poliuretana (TPU). Stabilna stanja odziva antagonistične fleksibilne NiTi SMA strukture so opazovali pri podaljšanem številu termičnih ciklov. Izdelali so primerjavo pri dveh različnih jakostih toka z reguliranim napajanjem z visokim enosmernim električnim tokom. Ciklični profil z maksimalno dvosmerno stabiliziranim aktuacijskim sunkom velikosti 52 mm so dosegli po stotih ogrevalno-ohlajevalnih ciklih. Avtorji so prav tako raziskovali vpliv dolžine ogrevalno-ohlajevalne dobe na stabiliziran odzivni sunek.

Ključne besede: zlitine z oblikovnim spominom, nikelj-titan, termične obremenitve, ciklično obremenjevanje

## 1 INTRODUCTION

Nickel-titanium shape-memory alloys are among the most widely used functional materials due to their outstanding shape memory and superelastic characteristics.<sup>1</sup> They reshape the designs and structural concepts of many engineering systems, such as biomedical instruments,<sup>2</sup> bio-inspired robots<sup>3</sup> and aerospace structures.<sup>4-5</sup> Particularly, NiTi SMA-actuated assembly systems are significant candidates to be used as elements embedded in flexible structures<sup>6</sup> due to their high power/weight ratio and compact housing volume. However, the stabilized response of an assembled system is required to be permanently used together with a consistent stroke or strength.

There are many competing methods for achieving a stable mechanical<sup>7</sup> or thermal<sup>8</sup> cyclic stroke. A combina-

tion of proper processing, structure and thermomechanical properties should be selected to achieve a repeatable functional performance of a NiTi SMA-based structure. The selected combination of these components determines the performance characteristics of the structure. Considering the processing methods, the chemical composition, the manufacturing technique and the heat treatment are the main parameters that affect the structure, the thermomechanical properties and, finally, the cyclic performance of an SMA-based actuator. The shape-memory effect and superelasticity are the two notable thermomechanical properties of the SMA materials. While the thermally induced martensitic transformation (TIMT) generates the shape-memory effect under isobaric conditions, the stress-induced martensitic transformation (SIMT) reveals the superelastic property under isothermal conditions. In terms of microstructure,

the shape-memory effect occurs with the reverse phase transformation from martensite to austenite. Self-accommodated martensite variants form through the TIMT. In contrast, favorably oriented martensite variants and further detwinning occur under mechanical loading during the SIMT. Upon unloading, the austenitic microstructure forms through reverse phase transformation.<sup>9</sup>

Most of the isothermal or isobaric experimental studies have focused on the feasibility of the linear or rotary actuation for single NiTi SMA wire<sup>10</sup> or spring<sup>11</sup> actuated structures. However, an antagonistically actuated SMA-based mechanism provides a larger actuation stroke with higher degrees of freedom compared to the single-SMA-element-actuated structures.<sup>12</sup> An antagonistic configuration consists of two opposing SMA elements, producing two-way mechanical work in an efficient manner. Many researchers worked on the antagonistic characteristics of the NiTi SMA, combining the shape memory and the superelastic responses using a wire, spring or plate. Sofla et al.<sup>13</sup> studied on the antagonistic SMA wires based structural morphing mechanism for the twisting segments of the wing. Williams and Elahinia<sup>14</sup> utilized a pair of antagonistic wires actuating the mechanism to replace the electrical motor-driven actuator in an automobile side mirror. A flap wing actuated by antagonistic NiTi SMA wires was examined by Senthilkumar<sup>15</sup> concentrating on the pulling force of the pair of the SMA wires. Recently, the thermomechanical response of the antagonistic NiTi SMA-based actuation has been studied using pairs of SMA wires,<sup>14,16–17</sup> springs<sup>18</sup> or plates.<sup>2,19</sup> Attempts have also been made to obtain a large stroke from SMA springs using a novel spring configuration.<sup>20</sup>

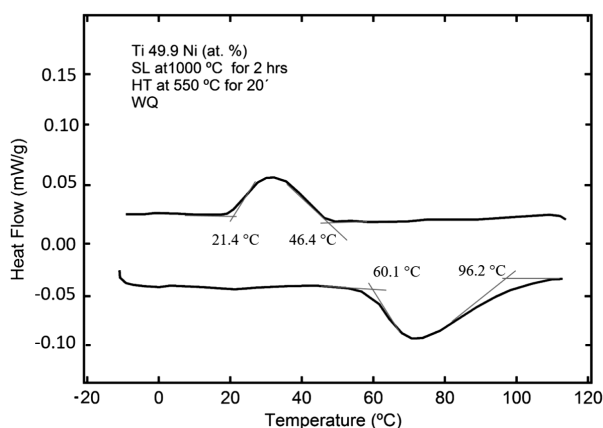
The NiTi SMA-based structures provide different benefits, such as self-healing,<sup>21</sup> self-sensing<sup>22–23</sup> and damping.<sup>24</sup> Attention must be paid to the stabilized response of the NiTi SMAs embedded in the structures under an extended number of thermal cycles. In this study, a novel flexible structure was developed using a couple of embedded, antagonistic NiTi SMA plates. The combination of processing-structure thermomechanical

properties were selected for the NiTi elements. To reveal the stabilized performance of the proposed NiTi SMA flexible structure, a prototype was manufactured and a special experimental set-up was established. The stabilized repetitive actuation stroke of the structure was investigated through 100 heating/cooling cycles under water condition.

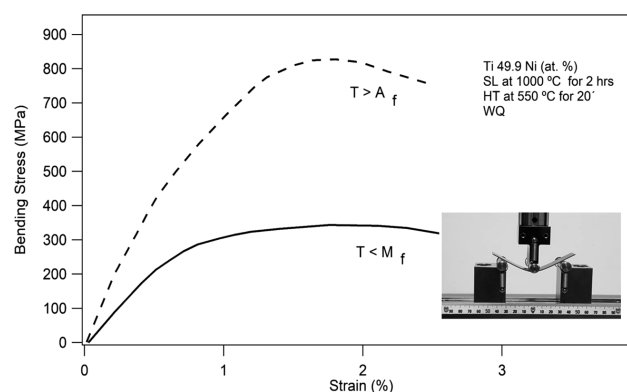
## 2 EXPERIMENTAL PART

The polycrystalline martensitic NiTi SMA plates with a thickness of 1 mm, produced by vacuum-induction casting and cold rolling were purchased from Memry (Weil am Rhein, Germany). The chemical compositions of NiTi ( $x_{Ti} = 49.9$  %) were determined using an X-ray fluorescence (XRF) analysis of the NiTi SMA plate samples. Differential scanning calorimetry (DSC) was utilized to determine the reverse and forward martensitic-phase-transformation temperatures of the plate samples. The DSC analysis was conducted at heating and cooling rates of  $15 \text{ }^\circ\text{C min}^{-1}$ . The phase-transformation temperature of the NiTi SMA sample measured with DSC is shown in **Figure 1**. The martensite and austenite start/finish temperatures were found to be  $M_f = 21.4 \text{ }^\circ\text{C}$ ,  $M_s = 46.4 \text{ }^\circ\text{C}$ ,  $A_s = 60.1 \text{ }^\circ\text{C}$  and  $A_f = 96.2 \text{ }^\circ\text{C}$ .

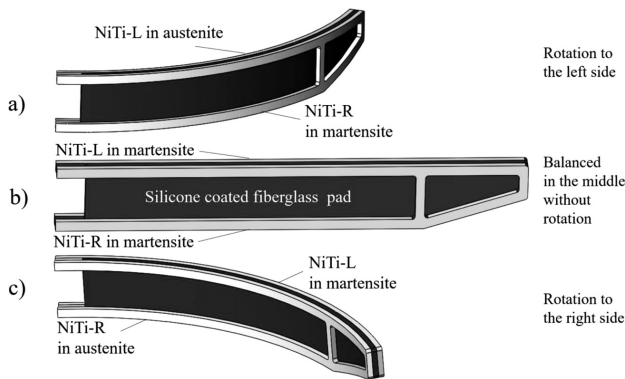
Three-point-bending (TPB) tests were conducted using a universal testing machine with a 5-kN load cell as shown in **Figure 2**. The NiTi SMA plate samples with dimension of  $(1 \times 100 \times 20)$  mm were used for the TPB tests at a 10 mm/min loading rate. The first test set was carried out at a temperature of  $21 \text{ }^\circ\text{C}$  and the NiTi SMA plate samples were in the martensitic state. Subsequently, the second test set was conducted with a DC power supply while the Joule heating of the NiTi SMA plate samples was above the  $A_f$  temperature, in a temperature range of  $110\text{--}130 \text{ }^\circ\text{C}$ . The three-point-bending test results for the NiTi SMA plates in fully martensitic and fully austenitic states are shown in **Figure 2**. The geometric design of the NiTi SMA plates was adapted to the design of the flexible mechanism. The NiTi SMA plates were cut using electrical discharge machining (EDM), following the final geometric design as shown in **Figure**



**Figure 1:** DSC results for the NiTi SMA plates



**Figure 2:** Three-point-bending test results for the NiTi plates in the martensitic and austenitic states

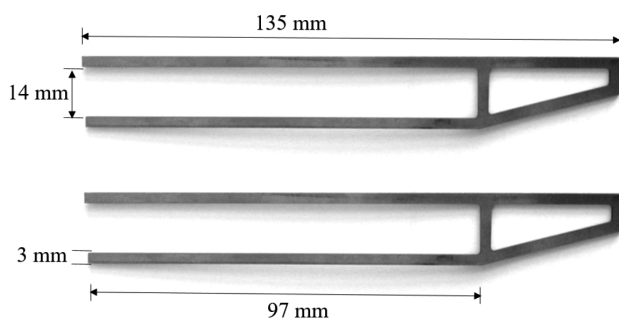


**Figure 3:** Design of the NiTi SMA plates: a) NiTi-L in the austenite phase / NiTi-R in the martensite phase, b) NiTi-L and NiTi-R in the martensite phase, c) NiTi-L in the martensite phase / NiTi-R in the austenite phase

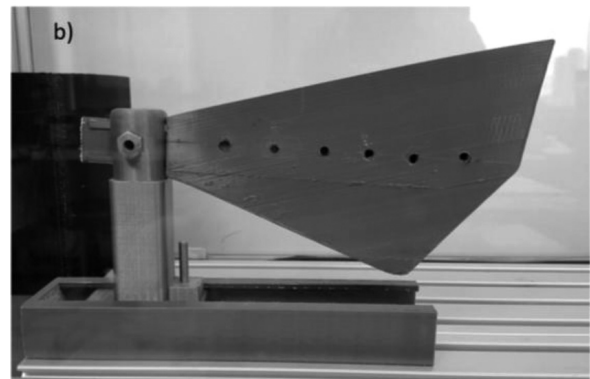
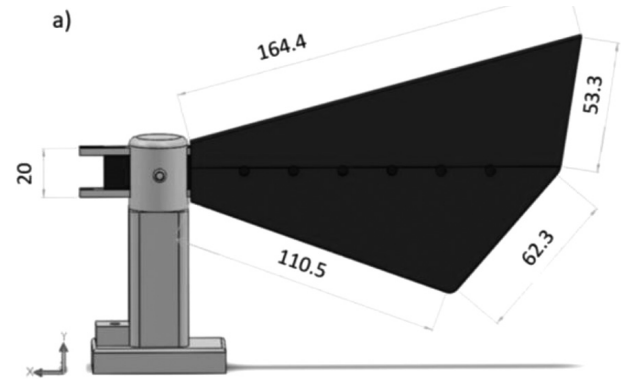
3b. The final plate geometry was designed to increase the area where the moment can be applied to the flexible wing system. The EDM-processed NiTi SMA plates were heat treated at 1000 °C for 2 h in an argon atmosphere in a vacuum furnace. The desired curved shapes, depicted in **Figures 3a** and **3c** as the austenitic form (NiTi-L and NiTi-R), were obtained with the heat treatment at 550 °C for 20 min and adapted to the designed aluminum mold.

After finishing the computer-aided design, the required flexible structure was built with additive manufacturing based on fused deposition modeling (FDM), using TPU filaments with a Shore A hardness of 85. The assembly components used for fixing the flexible structure to the experimental set-up were made of ABS (acrylonitrile butadiene styrene) filaments. The shape of the antagonistic NiTi SMA plates was geometrically adapted and embedded into the flexible structure. The final geometry of the electrical-discharge-machined (EDM) NiTi SMA plates is shown in **Figure 4**. The EDM-processed NiTi SMA plates were fixed antagonistically into the flexible structure for the rotational actuation. A flexible silicone-coated fiberglass pad was assembled between the two NiTi SMA plates for electrical insulation.

The design and prototype of the antagonistic NiTi SMA embedded in the flexible structure are shown in

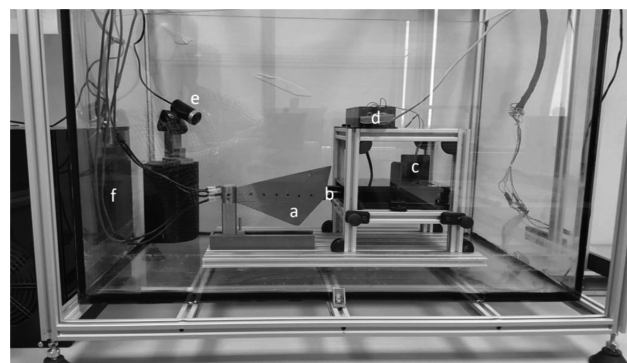


**Figure 4:** Final geometry of the electrical-discharge machined NiTi SMA plates



**Figure 5:** Design (a) and prototype (b) of the antagonistic NiTi SMA plates embedded in the flexible structure

**Figure 5.** Two different electrical-current levels of 60 A and 80 A were applied using a regulated high-current DC power supply. Two different heating/cooling cyclic periods were applied to observe the stabilized actuation. One is a high period; the other is a low period. In the high-period experiment, each of the antagonistic plates were heated alternately every 20 s. Specifically, each heating and cooling segment took place in 10 s, in which the electrical current was applied for 2 s. In the low-period experiment, each antagonistic plate was heated alternately for 2 s during an 8-second period. To examine the SMA-actuated flexible-structure characteristics under water conditions, an experimental set-up was established



**Figure 6:** Experimental set-up for the SMA-actuated flexible structure; a) NiTi SMA-actuated structure, b) load cell, c) infrared sensor, d) microcontroller, e) camera, f) DC power supply

as shown in **Figure 8**. An ABS fixture was fixed with screws onto an aluminum platform in a transparent water tank. An additively manufactured ABS joint was fixed to the ABS fixture. A silicone-coated fiberglass pad was placed between the NiTi SMA plates. The NiTi SMA plates were inserted inside the groove of the joint and fixed to the joint with two screws. The free side of the NiTi SMA plates was embedded into the additive-manufactured flexible enclosure. Heat-resistant electrical cables were connected to the legs of the NiTi SMA plates with heat-resistant electrical sockets.

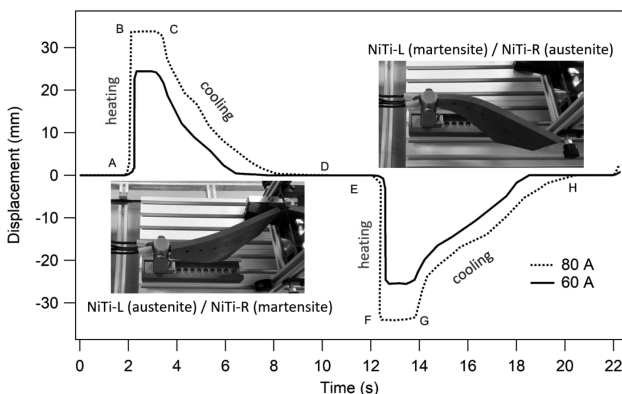
A load cell was placed at the trajectory of the tip of the flexible structure to measure the force applied to the SMA-actuated flexible structure as shown in **Figure 6**. A compression spring with a stiffness of  $0.65 \text{ Nmm}^{-1}$  was placed between the load cell and the blade tip. An infrared sensor was used to measure the actuation stroke at the tip of the flexible structure. Two thermocouples were placed on the surface of the NiTi SMA plates to measure the temperature change. Each thermal-cyclic experiment was conducted within the limits of  $21\text{--}98 \text{ }^\circ\text{C}$  including 100 thermal cycles. A water pump was used to keep the water temperature constant by circulating the water in the water tank.

### 3 RESULTS

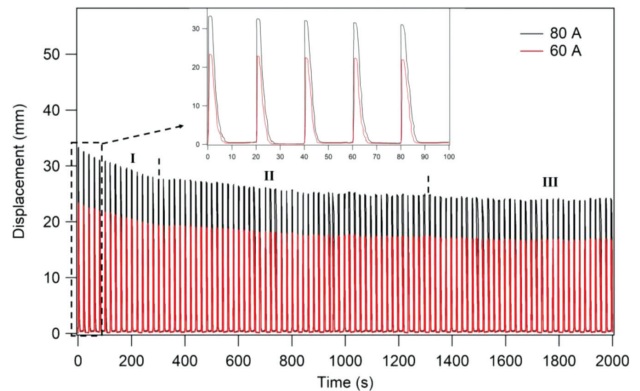
In order to obtain convenient Joule heating with a desired flexible-wing actuation, it is important to apply electrical currents to the antagonistic NiTi SMA plates without overheating them. Two different electrical-current values of 60 A and 80 A were used to observe the stabilized actuation during high and low heating/cooling cyclic periods. The displacement-versus-time results for each current level are shown in **Figure 5** for the high heating/cooling cyclic period. It is observed that the electrical current of 80 A created the maximum stroke value during the thermal loading. As the obtained actuation stroke depends on the applied heat above the austenite finish temperature ( $96.2 \text{ }^\circ\text{C}$ ), a larger stroke is

observed at 80 A compared to 60 A. Upon turning off the applied current, the temperature of the heated plate cools down to the martensitic phase. In this specific application, the martensite start temperature is  $46.4 \text{ }^\circ\text{C}$  as shown in **Figure 1**. A combined plot of the applied bidirectional stroke versus time responses for the two different electrical-current values is depicted in **Figure 8**. When the specified current flows through the NiTi SMA plate at the left side of the flexible structure (NiTi-L), this structure heats up until reaching a fully austenitic state. This segment is depicted as A, B and C in **Figure 7**. After a 2-second time interval (A to C), the cooling phase starts and continues between C, D and E. The martensitic-phase transformation is completed during this cooling period within 8 s. Upon heating the NiTi SMA plate at the right side (NiTi-R), a similar actuation occurs following the path of E-F-G-H as shown in **Figure 7**.

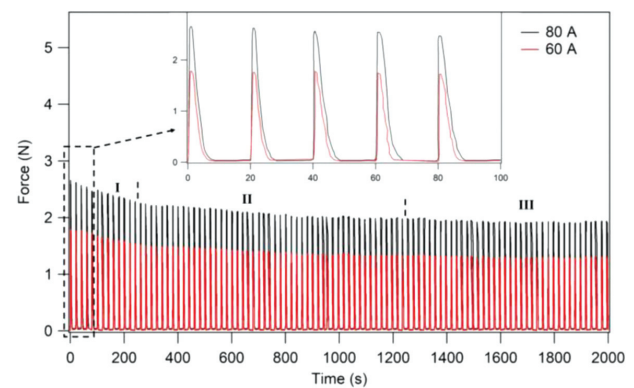
The displacement and force results, obtained for each current value are shown in **Figures 8 and 9**. It is noteworthy that a gradual decrease occurred during the thermal cycling. The cyclic-experiment results revealed that the slope of the decrease rate was consequentially divided into different stages including the early-evolu-



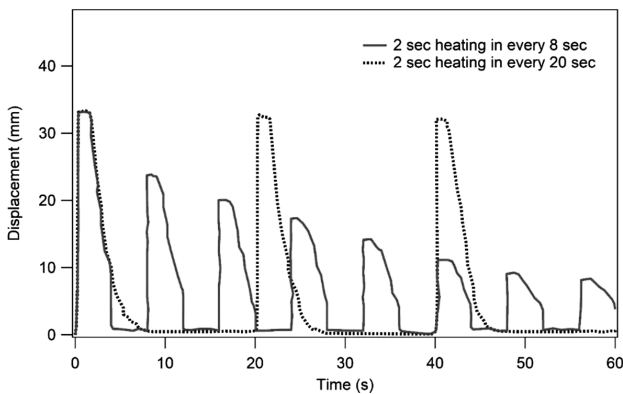
**Figure 7:** Antagonistic bidirectional displacement vs. time response of the flexible wing for the first thermal cycle with the electrical-current values of 60 A and 80 A



**Figure 8:** Displacement vs. time response of the embedded antagonistic NiTi SMA plate actuated flexible structure under 100 heating/cooling thermal cycles



**Figure 9:** Force vs. time response of the antagonistic NiTi SMA-actuated flexible structure under 100 heating/cooling thermal cycles



**Figure 10:** Comparison of the displacement vs. time responses of the antagonistic NiTi SMA-plate-actuated flexible structure with two different periods of heating/cooling thermal cycles

tion stage (I), transient stage (II, approach to the stabilization stage) and stabilization stage (III).

These stages were defined by three different slopes of the magnitudes of cyclic peaks for both the displacement and force data. We found that a notably large displacement occurred at 80 A compared to 60 A due to a faster Joule heating capacity. A stable actuation stroke of 26 mm was measured during the 100 heating/cooling cycles at 80 A. A total bidirectional maximum actuation stroke of 52 mm was reached through the antagonistic actuation when the electrical-current flow was 80 A. The early evolution occurred within the first 16 cycles as shown in **Figure 9**. Particularly, the displacement vs time plots for 80 A and 60 A start at 33 mm and 24 mm, ending at 26 mm and 17 mm. The cycles of the stages for 80 A and 60 A were 1–16 for the early evolution, 17–70 for the transient stage, and 71–100 for the stabilization stage.

Similar evolutionary characteristics were observed in the force vs time plots during the thermal cycling as shown in **Figure 9**. The transient stage started after 16 cycles and continued until the end of the 70<sup>th</sup> cycle. Finally, the stabilization stage was reached at the 71<sup>st</sup> cycle. Particularly, the force vs time plots for 80 A and 60 A start at 2.7 N and 1.7 N, ending at 2.1 N and 1.4 N.

A comparison was made between the actuation strokes obtained during high and low heating/cooling cyclic periods. The measured directional stroke decreased dramatically from 33 mm to 8 mm in the initial 8 cycles due to the low heating/cooling period (2-second heating in every 8-second period) as shown in **Figure 10**. The experimental results showed that a partial austenite/martensite transformation causes a fast degradation of the obtained stroke without reaching a proper stability. Contrary to the low heating/cooling period, the high heating/cooling period allows a stabilized actuation as shown in **Figures 8 and 10**.

## 4 DISCUSSION

The concept of an antagonistic NiTi SMA-actuated flexible structure is used to develop a compliant structure with a stabilized actuation. The geometrical shape of the antagonistic NiTi SMA plates is designed based on a flexible structure geometry. A systematic procedure is followed during the selection of the material required for the performance of the mechanism, considering the proper processing, structure and thermomechanical properties. To obtain the maximum stroke from the SMA embedded in the flexible structure, one of the antagonistic plates should be in the fully martensitic state, while the other plate transforms into the austenitic phase through Joule heating. There should be a synchronization of the phase transformation for both the NiTi-L and NiTi-R plates through cyclic Joule heating period. This synchronization covers the complete phase transformation. This procedure is crucial to prevent high degradation of the obtained stroke. Additionally, the bending stress in the martensite phase is lower than the bending stress in the austenite phase as shown in **Figure 2**. Thus, the heating/cooling thermal cyclic period of the antagonistic system should be designed according to these phenomena. When failing to obtain a fully martensitic phase during the cooling, a higher bending stress should be applied to bend the system in one direction. A comparison between the two different Joule heating periods clearly shows that a lack of sufficient water cooling greatly decreases the operational stroke of the flexible structure.

The cooling fluid and the antagonistic SMA configuration are two notable components, which accelerate the performance of the embedded-SMA-actuated structure. Typically, the response time of the SMA-actuated structure is fast upon heating. However, it takes too much time to cool down to the martensite phase, even with forced air convection. In recent studies, new cooling techniques have been investigated to increase the heat-transfer rate, such as air-jet-impingement cooling<sup>25–26</sup> and nanofluid cooling.<sup>27</sup> The proposed flexible structure is bidirectionally actuated by a couple of antagonistic SMA plates through alternate Joule heating under water conditions. The holes on the flexible structure increase the cooling rate of the antagonistic NiTi SMA plates, providing the contact with the water. A total of 10 s is required for each heating/cooling cycle for the proposed antagonistic structure. To obtain the maximum stroke from the SMA embedded in the flexible structure, one of the antagonistic plates should be in the fully martensitic state, while the other plate transforms into the austenitic phase through Joule heating. We note that the gained bidirectional stroke decreases dramatically when the heating/cooling cycle is not selected in accordance with the complete phase transformation. The three-point-bending test reveals that the bending strength of the austenite phase is much higher than the bending stress of the martensite phase. This explains that one of the

antagonistic plates needs a higher bending force to move the antagonistic NiTi SMA pair into one direction during Joule heating. Additionally, an early degradation of the gained stroke occurs.

Considering the design geometry, the antagonistic design can be selected in the forms of a wire, spring or plate configuration, depending on the application. However, the plate geometry of the antagonistic configuration provides various advantages in the design compared to the antagonistic wire design. The plate geometry can be transformed into different three-dimensional shapes. Moreover, it is easier to fix a plate component into any fixture than a wire counterpart. Additionally, an antagonistic pair of plates can apply the bending force in the opposite direction in a balanced manner compared to an antagonistic pair of wires.

The evolution of the thermal cyclic stabilization pattern should be taken into consideration to determine the long-term performance of the system. Any actuator structure needs to provide the required motion repeatability under service conditions. Thus, the major design requirement for a SMA-actuated flexible structure is a stabilized cyclic performance with a desired actuation stroke. When cyclic stability is reached, it can provide an enhanced sensorless position controllability of the flexible structure. An easy-to-use actuator control system is vital for an actuator. There has been a growing interest in the sensorless position control of the SMA-based actuator. Here, a stabilized large stroke obtained from an antagonistic SMA-actuated structure might be used for sensorless position control. Additionally, the excess number of the components, belonging to the actuator system affects the reliability of the system. To this end, the antagonistic actuator configuration presented in this study has the lowest number of components compared to the conventional actuator systems.

The experimental results show that the regulated high-current DC power supply with the maximum current value of 80 A is sufficient for the designed flexible-wing actuator system. We note that a lithium-ion rechargeable battery can be used to obtain a mobile mechanism considering the necessary power supply for the Joule heating. The potential application fields for the developed SMA-actuated flexible structure can be the steering components of mini autonomous underwater vehicles (AUV) or remotely operated underwater vehicles (ROV), such as a fin, rudder, wing, etc. Moreover, the flexible trailing edge and chord-wise bending are mainly used in aeronautics researches.<sup>5,13</sup> The developed antagonistic NiTi SMA-plate-actuated flexible-wing system can be adapted to different aerospace applications, such as small unmanned aerial vehicles (UAV) or satellite systems. To develop an advanced SMA-actuated mechanism for the above systems, an interdisciplinary study is required, covering material, mechanical, electrical and control subsystems.

## 5 CONCLUSIONS

In this study, a novel antagonistic NiTi SMA-actuated flexible structure was designed, built and evaluated under thermal cycling. A stable and repeatable actuation stroke was obtained from specific antagonistic NiTi plates embedded in the structure. The experimental results unveiled that the cyclic heating/cooling period should be selected based on the complete phase transformation of the antagonistic NiTi SMA plates synchronized through Joule heating. This synchronization is crucial for preventing a notable cyclic degradation of the gained stroke from an antagonistic SMA plate configuration. Additionally, the cyclic stroke tended to stabilize through thermal cycling during three remarkable stabilizing stages. These are the early-evolution stage, transient stage and stabilization stage. The early evolution occurred in the initial 16 cycles. The transient stage took place between the 17<sup>th</sup> and 70<sup>th</sup> cycles. Finally, the stabilization stage was reached after the 71<sup>st</sup> cycle. The largest stable bidirectional actuation stroke of 52 mm was obtained for this specific antagonistic design. In the future, the developed antagonistic NiTi SMA-actuated flexible structure will be used for a bio-inspired underwater application.

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